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Research paper



Periodic orbit in the photo-gravitational restricted three body problem around the collinear Lagrangian points when more massive primary is an oblate spheroid and source of radiation

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Abstract

The circular Restricted Three Body Problem is considered with the more massive primary as an oblate spheroid and source of radiation. A new mean motion expression given by n2=1+6A is used in the present study, when the secular effect of the oblateness on the mean motion, argument of perigee and right ascension of the ascending node is considered. The locations of the collinear Lagrangian points are found. It is found to have some variation from the previous study conducted on the same because of the new mean motion that is considered in this study. The variations of the location of the Lagrangian points due to the unperturbed as well as the perturbed problem due to oblateness and radiation pressure are studied. A study on the eccentricity e and angular frequency s at L1, L2 and L3 is carried out. It is observed how the change in effect of oblateness and radiation pressure has affected the location, angular frequency and eccentricity at L3, though there are only small changes noticed in the case of L1 and L2.

Keywords: Angular Frequency; Collinear Lagrangian Points; Oblateness; Periodic Orbits; Radiation Pressure; Restricted Three Body Problem.

1. Introduction

The restricted three-body problem describes the motion of an infinitesimal mass moving under the gravitational effect of the two finite masses, called primaries, which move in circular orbits around their center of mass on account of their mutual attraction and the infinitesimal mass not influencing the motion of the primaries. The RTBP (restricted three-body problem) possesses five stationary solutions called Lagrangian points, three of which are called collinear equilibria, lie on the line joining the primaries and the other two called equilateral equilibria make equilateral triangles with primaries. In general, the collinear equilibria are unstable while equilateral points are stable only in a certain region for the mass parameter. The shapes of the participating bodies in the RTBP are assumed to be oblate. The oblateness of a body can produce perturbation deviation from the two-body motion. In the RTBP it is assumed that the infinitesimal mass moves under the mutual gravitational force of the primaries. In RTBP when at least one of the interacting bodies is an intense emitter of radiation the repulsive force of the radiation pressure is considered. In the RTBP the origin is considered to be the center of mass of the two finite mass bodies and take axes rotating with the masses, such that they lie along the x-axis. In the dimensionless system of variables μ_1 and μ_2 represents the masses of the primaries and the sum of two masses is therefore unity and whose distance are also unity. In this study the larger mass is taken on the right side. The book of Szebehely (1967) provides systematic coverage of the literature on the subject as well as derivations of some of the important results.

Various authors have made studies on Lagrangian points in the restricted three-body problem by considering the more massive primary or both primaries as source of radiation. Some of the important contributions are by Radzievsky (1950, 1953), Chernikov (1970). Some of the significant studies carried out related to the Lagrangian points by considering the oblateness of one or both the primaries with their equatorial planes coincident with the plane of motion, are by Vidyakin (1974), Sharma (1975), Subba Rao & Sharma (1975), Sharma & Subba Rao (1975, 1976, 1978).

In this paper we study the RTBP when the more massive primary is a source of radiation as well as an oblate spheroid. A study on the same had been done by (Namboodiri et al. 2008) but with the mean motion value of 1+3A/2. In this paper when the secular effect of the oblateness on the mean motion, argument of perigee and right ascension of the ascending node is considered the mean motion is taken to be 1+6A derived analytically by (Sharma et al. 2020). We have derived the expressions for locations of the collinear points and a numerical study on the effect of oblateness and radiation pressure is carried out. We have also carried out study on the effect of oblateness and radiation pressure on the angular frequency and eccentricity with new mean motion.



2. Equation of motion

In the dimensionless synodic coordinate system (x, y), the equations of motion are (Szebehely 1967)

$$\ddot{\mathbf{x}} - 2\mathbf{n}\dot{\mathbf{y}} = \frac{\partial\Omega}{\partial \mathbf{x}},$$
$$\ddot{\mathbf{y}} + 2\mathbf{n}\dot{\mathbf{x}} = \frac{\partial\Omega}{\partial \mathbf{y}},$$
(1)

Where

$$\Omega = \frac{n^2}{2} \left[(1-\mu)r_1^2 + \mu r_2^2 \right] + \frac{q(1-\mu)}{r_1} + \frac{\mu}{r_2} + \frac{q(1-\mu)A}{2r_1^3}.$$
(2)

With $r_1^2 = (x - \mu)^2 + y^2$, $r_2^2 = (x + 1 - \mu)^2 + y^2$, $n^2=1+6A$.

The mass parameter $\mu = m_1 / (m_1 + m_2)$, where m_1 and m_2 are the masses of the primaries $m_1 > m_2$, such that $m_1 + m_2 = 1$ the oblateness coefficient $A = \frac{AE^2 - AP^2}{5R^2}$, where AE and AP are the dimensional equatorial and polar radii of the more massive primary and R is the distance between the primaries. The term q is due to radiation pressure. If q =1, then there is no effect of radiation pressure. For equilibrium points $\Omega_x = \Omega_y = 0$ Therefore

$$n^{2}x - \frac{q(1-\mu)(x-\mu)}{r_{1}^{3}} - \frac{\mu(x-\mu+1)}{r_{2}^{3}} - \frac{3Aq(1-\mu)(x-\mu)}{2r_{1}^{5}} = 0,$$
(3)

$$y\left[n^2 - \frac{q(1-u)}{r_1^3} - \frac{u}{r_2^3} - \frac{3Aq(1-u)}{2r_1^5}\right] = 0.$$
(4)

3. Location of collinear equilibrium points

When y = 0, eq. (collinear points $L_1(x_1,0)$, $L_2(x_2,0)$, $L_3(x_3,0)$, where (3) determines the locations of the

$$x_1 = \mu - 1 - \xi_1, x_2 = \mu - 1 + \xi_2, x_3 = \mu + \xi_3.$$

 ξ_1, ξ_2, ξ_3 satisfies the seventh-degree polynomials

 $(12A+2)\xi_{1}^{7} + [(-12A-2)\mu + 60A + 10]\xi_{1}^{6} + [(-48A-8)\mu + 120A + 20]\xi_{1}^{5} + [(2q-72A-14)\mu - 2q + 120A + 20]\xi_{1}^{4} + [(4q-48A-16)\mu - 4q + 60A + 10]\xi_{1}^{3} + [[(3A+2)q - 12A - 14]\mu + (-3A-2)q + 12A + 2]\xi_{1}^{2} - 8\mu\xi_{1} - 2\mu = 0,$ (6)

 $(12A + 2)\xi_2^7 + [(12A + 2)\mu - 60A - 10]\xi_2^6 + [(-48A - 8)\mu + 120A + 20]\xi_2^5 + [(-2q + 72A + 10)\mu + 2q - 120A - 20]\xi_2^4 + [(4q - 48A)\mu - 4q + 60A + 10]\xi_2^3 + [[(-3A - 2)q + 12A - 10]\mu + (3A + 2)q - 12A - 2]\xi_2^2 + 8\mu\xi_2 - 2\mu = 0,$ (7)

 $(12A + 2)\xi_3^7 + [(12A + 2)\mu + 24A + 4]\xi_3^6 + [(24A + 4)\mu + 12A + 2]\xi_3^5 + [(2q + 12A)\mu - 2q]\xi_3^4 + (4q\mu - 4q)\xi_3^3 + [(3A + 2)q\mu + (-3A - 2)q]\xi_3^2 + (6Aq\mu - 6Aq)\xi_3 + 3Aq\mu - 3Aq = 0.$ (8)



Fig. 1: Graph Showing the Variation of Location of L1 versus μ .

(5)



Fig. 3: Graph Showing the Variation of Location of L3 versus μ .

Fig, 1, 2 and 3 provides the locations of the collinear points L_1 , L_2 and L_3 for μ from 0 to 0.5 for different values of A_1 and q. In Fig. 1it is noted that the collinear point L_1 comes nearer to the primaries with the increase in oblateness and radiation pressure, and also it is noted that at smaller values of mass parameter μ , the location moves away from the primaries, but moves towards the primaries for the values of μ above 0.2. In Fig. 2 it is seen that L_2 move away from the more massive primary with the increase in oblateness and come nearer to it with the increase in radiation pressure. Fig. 3 shows that L_3 moves closer to the primaries with the increase in oblateness and radiation pressure.

4. Variational equations and characteristic exponents

The variational equations in the linear analysis becomes

$$\xi'' - 2n\eta' = \Omega_{xx}(a,b)\xi + \Omega_{xy}(a,b)\eta,$$

$$\eta' + 2n\xi' = \Omega_{xy}(a, b)\xi + \Omega_{yy}(a, b)\eta.$$

The characteristic equation of (9) is given by

$$\lambda^4 + (4n^2 - \Omega_{xx} - \Omega_{yy})\lambda^2 + \Omega_{xx}\Omega_{yy} - \Omega_{xy}^2 = 0.$$

At the collinear points, we get

$$\Omega_{xx} = n^2 + \frac{2q(1-\mu)}{r_1^3} + \frac{2\mu}{r_2^3} + \frac{6q(1-\mu)A}{r_1^5} > 0,$$

$$\Omega_{xy} = 0,$$

$$\Omega_{yy} = n^2 - \frac{q(1-\mu)}{r_1^3} - \frac{\mu}{r_2^3} - \frac{3q(1-\mu)A}{2r_1^5} < 0.$$

Consequently, $\Omega_{xx}\Omega_{yy} - \Omega_{xy}^2 < 0$. We can note that the roots $\lambda_i (i = 1,2,3,4)$ of the characteristic equation (10) are

(9)

(10)

$$\lambda_{1,2} = \pm \left[-\beta_1 + (\beta_1^2 + \beta_2^2)^{\frac{1}{2}} \right]^{\frac{1}{2}} = \pm \lambda,$$

$$\lambda_{3,4} = \pm \left[-\beta_1 - (\beta_1^2 + \beta_2^2)^{\frac{1}{2}} \right]^{\frac{1}{2}} = \pm is,$$

Where

$$\beta_1 = 2 - \frac{\Omega_{xx} + \Omega_{yy}}{2},$$

$$\beta_2^2 = -\Omega_{xx}\Omega_{yy} > 0.$$

It is noted that $\lambda_{1,2}$ are real and $\lambda_{3,4}$ are pure imaginary. Hence, in general case the collinear equilibria are unstable. The eccentricity of the periodic orbit is given by $e = \sqrt{1 - \beta_3^{-2}}$.

Where

 $\beta_3 = (s^2 + \Omega_{xx})/2ns.$





Fig. 4, 5 and 6 provide the angular frequencies s_1 , s_2 and s_3 at L_1 , L_2 and L_3 , respectively for the mass parameter μ up to 0.5 for different values of A_1 and q. It may be noted from Fig. 4 that the increase in the effect of oblateness and radiation pressure increases s_1 values for L_1 . At lower values of μ the difference between the angular frequencies for radiation pressure is high but at higher values of μ the difference is less. From Fig. 5 it may be noted that the angular frequency s_2 at L_2 increases with oblateness and decreases with increase in radiation pressure q. At lower values of μ the increase in s_2 is noticeable but as the μ value increases the variation in s_2 decreases. In Fig. 6 it is seen that, at L_3 the increase in oblateness and radiation pressure increases s_3 .



Fig. 8: Graph Showing the Variation of Eccentricity at L2.



Fig. 9: Graph Showing the Variation of Eccentricity at L3.

Fig. 7, 8 and 9 provide the eccentricity e_1 , e_2 and e_3 at L_1 , L_2 and L_3 for the mass parameter μ up to 0.5 for different values of A_1 and q. It may be noted from Fig. 7 that at L_1 the increase in oblateness and radiation pressure increases eccentricity, also as μ increases the eccentricity decreases. It is seen from Fig. 8 that for L_2 eccentricity increases with oblateness and decreases with increase in radiation pressure, also at $\mu < 0.1$ the increase in eccentricity is very high but becomes almost constant for values of μ above 0.25. From Fig. 9 at L_3 the eccentricity increases with oblateness and radiation pressure and the increase become linear at $\mu > 0.15$.

5. Conclusion

A study on the effect of oblateness and radiation pressure on the location, angular frequency and the eccentricity of the conditional periodic orbits at L_1 , L_2 and L_3 is carried out with the new mean motion $n^2 = 1 + 6A$. The present study is different from the previous study conducted on the same with mean motion $n^2 = 1+3A/2$ by (Namboodiri et al. 2008). In the current study the mean motion includes the secular effect of the oblateness on the mean motion, argument of perigee and right ascension of the ascending node. The effect of this new mean motion can be seen when the two studies are compared. It can be noticed how the change in effect of oblateness and radiation pressure has affected the location, angular frequency and eccentricity at L_3 , though the changes at L_1 and L_2 due to the effects of oblateness and radiation is barely noticeable.

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